

Application No. 10/628,556  
Amendment After Allowance dated March 20, 2008

**Amendments to the Claims:**

This listing of claims will replace all prior versions and listings of claims in the application.

**Listing of Claims**

1. (Currently Amended) A casing for a non-expendable turbofan engine, substantially encasing at least a fan assembly, a compressor assembly, a combustor assembly and a turbine assembly, the casing comprising:

a fan case portion surrounding the fan assembly;

an annular intermediate case portion having an annular outer portion aligned with  
and projecting axially rearwardly from the fan case portion and an annular  
inner portion; and

a gas generator case portion aligned with and extending axially rearwardly from the  
intermediate case annular inner portion and housing the combustor assembly,

wherein the fan case portion, the intermediate case annular outer and inner portions and the  
gas generator case portion are integrally joined together, thereby forming an integral casing.

2. (Original) The casing as claimed in claim 1 wherein the fan case portion, the  
intermediate case portion and the gas generator case portion are made of the same material.

3. (Original) The casing as claimed in claim 1 wherein the intermediate case  
portion further comprises an integral compressor shroud portion and an integral bearing  
mount portion.

4. (Cancelled)

5. (Previously amended) The casing as claimed in claim 1 wherein a first weld joins the  
fan case portion to the intermediate case portion, and wherein a second weld joins the gas  
generator case portion to the intermediate case portion.

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6. (Original) The casing as claimed in claim 1 wherein the fan case portion, the intermediate case portion and the gas generator case portion are joined together by flangeless connections.

7 to 16. (Cancelled)

17. (Previously presented) The casing as claimed in claim 1, wherein the intermediate case portion comprises a splitter including inner and outer annular walls extending axially rearwardly from a leading edge tip, the outer annular wall forming a radially inner boundary of a bypass air flow, the inner annular wall forming a radially outer boundary of a main gas path of the turbofan engine.

18. (Previously presented) The casing as claimed in claim 1, wherein the fan case portion has an inner shroud which is integrally connected to an inner hub of the intermediate case portion.

19. (Previously presented) A non-expendable turbofan engine comprising a casing as defined in claim 1.